### UNCLASSIFIED

AD 272 123

Reproduced by the

ARMED SERVICES TECHNICAL INFORMATION AGENCY
ARLINGTON HALL STATION
ARLINGTON 12, VIRGINIA



UNCLASSIFIED

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

272123 272123

### MELLON INSTITUTE

Quarterly Scientific Progress Report
Report No. 15
for the period
November 1, 1961 - January 31, 1962

### STRESS CORROSION OF HIGH STRENGTH STEELS AND ALLOYS; ARTIFICIAL ENVIRONMENT

Research Project No. 389-2



Sponsored by
U. S. Army Ordnance, Frankford Arsenal
Mr. H. Rosenthal, Contract Monitor

Contract No. DA-36-034-ORD-3277RD

NEROX

DOIA OO

Dr. H. L. Anthony Staff Adviser

### **MELLON INSTITUTE**

Report No. 15

### QUARTERLY SCIENTIFIC PROGRESS REPORT

for the period

November I, 1961 - January 31, 1962

### STRESS CORROSION OF HIGH STRENGTH STEELS AND ALLOYS; ARTIFICIAL ENVIRONMENT

Research Project No. 389-2

Sponsored by
U. S. Army Ordnance, Frankford Arsenal
Mr. H. Rosenthal, Contract Monitor

Contract No. DA-36-034-ORD-3277RD

This Report Presented By The Project Staff

C. J. Owen, Fellow W. D. Ruble, Research Assistant

The publication of this report does not necessarily constitute approval by the Army of the findings or conclusions contained therein.

### TABLE OF CONTENTS

		Page
	ABSTRACT	ii
I.	INTRODUCTION	1
II.	EXPERIMENTAL PROCEDURES	2
III.	EXPERIMENTAL	7
	A. Sample Preparation and Heat Treatment	7
	B. Stress Corrosion Testing	8
	1. Bent Beam Tests	8
	a. D6Ac	8
	b. Vascojet 1000	9
	c. AM355	10
	d. PH15-7 Mo	11
	e. B120VCA	11
	f. 4137 Co	11
	2. U-Bend Tests	12
	a. 300M	12
	b. AM355	13
	c. PHI 5-7 Mo	14
	d. B120VCA	14
	e. 4137 Co	15
	f. Ardeform 301	16
	g. U-modified 4137 Co	16
T3 <i>t</i>	FIITIDE WORK	1Ω

### **ABSTRACT**

A brief review of experimental procedures, including test methods, apparatus, sample preparation, and testing is presented for reference purposes.

Cumulative stress corrosion test data for the various bent beam and U-bend tests on high strength missile alloys are given.

Included are bent beam data for D6Ac, Vascojet 1000, AM355, PH15-7

Mo, B120VCA, and 4137 Co and U-bend data for 300M, AM355, PH15-7

Mo, B120VCA, 413? Co, Ardeform 301, and U-modified 4137 Co.

While testing is not as yet complete and has not progressed to a sufficient degree in all cases for adequate summation, the low alloy steels 4137 Co, 300M, and Vascojet 1000 are definitely susceptible to stress corrosion in varying degree. The AM355, PH15-7 Mo, Bl20VCA, and Ardeform 301 alloys are not susceptible, regardless of strength level of exposure.

Alloy modification of the 4137 Co with uranium has very markedly decreased the stress corrosion susceptibility of this alloy.

### I. INTRODUCTION

The project work described herein represents a portion of a grant made available by the Army to promote a general scientific advancement in the area of case materials for missiles. This specific project is concerned with the synthetic environment stress corrosion testing of specified high strength steels and alloys. The research objectives of the project were presented in Report No. 1.

Natural environmental tests on high strength steels and alloys are being conducted by Aerojet General Corporation, with actual production environments and rocket propellant environments being utilized. By prior mutual agreement, the same steel and alloy sheet material will be used for both projects and possible heat treatment variations will be circumvented by exchanging heat treated material whenever possible.

For the tests described herein, bent beam and U-bend sample; were cut and machined from available sheet material and subjected to stress corrosion testing in five selected synthetic environments in the laboratory.

This report presents further information on continuing
U-bend and bent beam tests for the assigned alloys, and for Rocoloy
270, 4137 Co, and Ardeform 301 missile steels.

### II. EXPERIMENTAL PROCEDURES

### Test Methods

A discussion of the U-bend and bent beam test methods together with an outline of the synthetic stress corrosion test environments used in performing the research are given in the July, 1960, Monthly Scientific Report, Report No. 1.

All recently-exposed (and future) samples being subjected to stress corrosion testing are being weighed prior to exposure to the test solutions to provide, whenever possible, general corrosion information.

### Apparatus

Schematic drawings of bent beam sample holders, the
U-bend test holders, a sample bending device for bent beam specimens,
and a stress corrosion test tank were presented in Report Nos. 1 and 3.

Polyethylene containers are presently being used for the stress corrosion exposure of U-bend specimens to the various synthetic environments. Each container will adequately hold six U-bend specimens. The use of these containers will supplement the sample testing presently being conducted in epoxy-coated tanks. Bent beam specimens are being tested in heavy glass dishes and epoxy-coated metal tanks.

Construction of shelving and an aeration system (Report No. 6) together with other pertinent items has facilitated (wherever feasible) the transfer of some samples under test from the glass containers to the epoxy-coated test tanks. The completed facility is illustrated in Report No. 7.

### Alloy Sample Material

The alloys being used in the study on stress corrosion include:

- 1. Low Alloy: Ladish D6Ac
- 2. Si-Modified 4300 Series: 300M.
- 3. Hot-Worked Die Steel: Vascojet 1000.
- 4. Cold-Worked PH Steel: AM355.
- 5. Heat-Treated PH Steel: PHI5-7 Mo.
- 6. Titanium Alloy: Bl20VCA.
- 7. Low Alloy-Cobalt Modified: 4137 Co.
- 8. Low Alloy-Cobalt Modified: Rocoloy 270.
- 9. Stretch-Formed 17/7 Stainless Steel: Ardeform 301.
- 10. Low Alloy-Co and U-Modified: 4137 Co.

All of the sample material is being tested in sheet form and was procured as such.

Chemical analyses for the above listed alloys and for the comparative heats are given in their entirety in Table I in Report

No. 13. Physical properties and heat treating surveys for the alloys were presented in Report No. 12.

### Sample Preparation

Sheet material of the various alloys was sheared into slightly over-size U-bend, bent beam, and tensile specimens in both the primary (direction of lowest Y.S.) and secondary directions.

The specimens were then wet-ground to width, length, and thickness (if necessary). Sample thickness was maintained between 30 and 50 mils, and as close to 50 mils as possible.

After heat treatment surveys were conducted (see Report No. 12) to determine the Y.S. variation with tempering temperature for each alloy (not necessary for AM355), the specimens were heat treated to strength levels approximating 200, 220, 240 Kpsi and maximum Y.S. Heat treating of the Bl20VCA alloy consisted of aging in an inert atmosphere (see Report No. 12).

All the surfaces except the ends of the heat treated specimens were then polished with emery paper to remove oxide films, the final polish being administered with 240-grit emery. Care was taken to assure a uniformly square edge on the specimens. A tendency of the D6Ac, V-1000, and 300M specimens to decarburize was circumvented by carefully wet-grinding the surfaces of the heat-treated samples an additional 5 mils per surface.

Tensile and metallographic samples were also cut from the alloy sheet material for the determination of as-received and heat-treated physical properties.

### Sample Testing

The U-bend specimens were bent into a V-shape with a special bending apparatus, sprung into clips (see Report No. 12), and immersed in each of the five test solutions (one-molar NaCl, Na<sub>2</sub>SO<sub>4</sub>, NaNO<sub>3</sub>, NaPO<sub>3</sub>, and Na<sub>2</sub>S).

The bent beam specimens were precision-cut to length

(+ 0.005 inches) and sprung into a holder with a bending device (see

Report No. 12). Immersion in the above five solutions constituted the
test phase.

### Status of Sample Material

The present status of the alloy sample material to date:

- D6Ac heat treatment complete; some bent beam samples being tested.
- 300M heat treatment complete; some U-bend data available.
- 3. Vascojet 1000 heat treatment complete; some bent beam data available.
- 4. AM355 all U-bends and all bent beam samples under test.

- 5. PHI5-7 Mo all U-bends and some bent beam samples under test.
- 6. Bl20VCA heat treatment complete; some U-bend and bent beam samples under test.
- 7. 4137 Co various U-bends under test.
- 8. Rocoloy 270 undergoing final machining.
- 9. Ardeform 301 U-bends under test.

### III. EXPERIMENTAL

Sample preparation, heat treatment, metallographic examination, and stress corrosion testing of all alloys, both assigned materials and additional samples accepted for study, are continuing.

### A. Sample Preparation and Heat Treatment

During the period of this report, the 300M samples were heat treated to three yield strength levels of about 200, 220, and 250 Kpsi. The heat treatment schedule was in accordance with the information given in the heat treatment survey (see Report No. 12) for this alloy.

U-bend samples subsequent to heat treating and precision cutting to length of bent beam samples constituted the greater portion of the work during this quarter. Wet grinding and emery polishing with 240-grit emery cloth as the final step, were utilized in removing oxide films from the samples. The bent beam samples were cut to length on a cut-off wheel with water coolant to the lengths required. Length calculation for stressing to 75% of the various yield strength levels was accomplished by the Phelps and Loginow method. \*

<sup>\*</sup> Phelps, E. H., and Loginow, A. W., "Stress Corrosion of Steels for Aircraft and Missiles," Corrosion 16, July (1960), 97-107.

Tabular data pertinent to the length calculation was supplied by Dr. Phelps and was utilized in this work.

Sample preparation for the investigation of coatings and their contribution to alleviation of stress corrosion failure has been initiated. Since surface tension effects associated with any dip-coating composition will tend to leave an uncoated edge if that edge is square, all these samples will have rounded edges for coating and testing.

Uncoated control samples of these alloys will be run simultaneously for comparative purposes.

### B. Stress Corrosion Testing

During this report period, a number of Vascojet 1000, D6Ac, AM355, PH15-7Mo, and Bl20VCA bent beam samples were subjected to synthetic environment stress corrosion testing. U-bend specimens of 300M and a uranium-modified 4137 Co alloy were also subjected to testing during this period. A more detailed discussion of the results to date for these alloys is provided in the following sections of this report.

### 1. Bent Beam Tests

Bent beam specimens of all assigned alloys except 300M are presently undergoing stress corrosion testing. These include:

a. DóAc - Only the intermediate strength bent beam samples for this alloy have been subjected to stress corrosion testing to date. The cumulative results, as illustrated in Table I, indicate that this alloy may prove to be superior in stress corrosion resistance to the other low alloy, high-strength steels (V-1000, 300M, and 4137 Co).

b. Vascojet 1000 - Data for bent beam specimens from all three yield strength levels are available for this alloy. Data obtained from exposure of primary direction (transverse) samples are given for the low strength level (~205 Kpsi) in Table II, for the intermediate strength level (~245 Kpsi) in Table III, and for the secondary direction (longitudinal), high strength level (~260 Kpsi) in Table IV.

Even though the results are limited essentially to one direction per strength level at this time, we can conclude the alloy is susceptible to stress corrosion failure at the high and intermediate yield strength levels in all but the sulfide environment. In the chloride environment it is apparently susceptible at all strength levels. The reason for the disparity in failure time in nitrate solution between the intermediate (Table III) and high (Table IV) strength levels is not immediately apparent.

To date, the low strength (Table II) specimens have not shown a susceptibility to stress corrosion in the nitrate, sulfide, sulfate, or phosphate solutions.

Table IV also provides a comparison between 4-inch and 7-inch specimens, secondary direction, in chloride and nitrate solutions. The results are of the same order of magnitude in both average time to failure and failure time range. We may conclude from this that specimen length is of little consequence if the  $\frac{C}{E}$  versus  $\frac{L-H}{H}$  point calculated for the specimen falls on the section of greater slope of the H/t curve shown in Figure 2 of the Phelps-Loginow publication.

c. AM355 - Although the primary (transverse)

direction specimens (Table V) have been under test for only a

relatively short period, the secondary direction (longitudinal or

rolling direction) samples (Table VI) have undergone exposure for

a considerable period. No failures to date have been recorded at

any of the three yield strength levels in the five synthetic environ
ments. For all practical purposes, this alloy material is not

susceptible to stress corrosion failure in the environments utilized.

The Table VI specimens were removed from their holders after approximately 300 days of environmental testing so that the holders could be utilized for other samples. Some permanent set was noted in the samples after removal. This is not surprising, considering the shape of the tensile curve for this alloy, in that there is little or no linear relationship between stress and strain

below the 0.2% offset Y.S. The tensile curve describes a sweeping arc to fracture.

d. PHI5-7 Mo - Samples from only one direction at the highest attainable yield strength (~290 Kpsi) have been exposed to stress corrosion testing to date. There have been no failures, however, in any test environment, as indicated in Table VII. We can conclude, on a preliminary basis, that the alloy is not prone to stress corrosion.

e. Bl20VCA - Preliminary data for one direction at the high strength level (~200 Kpsi) are given in Table VIII. No failures have been recorded to date and the alloy does not seem to be stress corrosion sensitive.

f. 4137 Co - There has been no change during this report period of the status of the 4137 Co bent beam samples with the exception of one sample failure in the 75% of Y.S. group, 175 Kpsi applied stress, severe industrial exposure.

A discussion of the results obtained from these natural environment exposures has been presented in Report Nos. 12, 13, and 14. The cumulative data for the three environments are presented in Tables IX and X for reference purposes.

### 2. U-bend Tests

During this report period, a number of U-bend samples of 300M, AM355, and U-modified 4137 Co have been exposed to the various synthetic test environments. The cumulative data to date for all alloys under test are discussed in the following sections of this report.

a. 300M - A number of 300M U-bend specimens of both the Primary Heat and Comparative Heat sheet materials were subjected to stress corrosion testing in the five synthetic environments. All samples from the low (~200 Kpsi Y.S.), intermediate (~220 Kpsi Y.S.), and high strength (~250 Kpsi Y.S.) groups for both directions (primary and secondary) have not been exposed to the test solutions. However, the cumulative results to date are given in Tables XI, XII, XIII, and XIV.

On a preliminary basis, the alloy shows a susceptibility to stress corrosion and the results are of approximately the same order of magnitude as those obtained for the 4137 Co alloy. The low (Table XI) and intermediate (Table XII) strength samples were unaffected in all but the phosphate solution, in which all samples failed in minutes or hours. The exposure time has been too short to permit further comparison.

Failure time for the high strength 300M samples (Tables XIV and XV) was of the order of minutes or hours in all except the sulfide environment, in which no failures were recorded for any strength level. The agreement among the times to failure of the Primary Heat and Comparative Heat high strength samples was not too good, indicating a possible considerable variation from heat to heat for this alloy.

b. AM355 - The cumulative U-bend test results for this alloy are presented in Tables XV (low strength), XVI (intermediate strength), and XVII (high strength). Sample failures have only been recorded in the chloride test solution and more predominantly in the primary direction samples.

The possibility that the samples were subjected to tensile-surface cracking during bending was investigated for the low strength (Table XV), primary direction samples prior to immersion during this report period. Surface cracking was observed under 10-power and 30-power magnification and confirmed by cross-sectioning, polishing, and by subsequent metallographic examination. Since the bent beam AM355 samples (Tables V and VI) did not exhibit stress corrosion sensitivity, we can conclude that, while this alloy is not stress-corrosion prone even in the stressed (below the Y.S.) state, it is subject to possible failure in plastically-deformed areas

in a chloride environment. The crack size necessary for susceptibility is problematical.

c. PHI5-7 Mo - The cumulative data for the stress corrosion testing of PHI5-7 Mo U-bends indicate that, after from 100 to 300 days of environmental exposure, the alloy is definitely not susceptible to stress corrosion failure. Not a single sample failure was recorded for any strength level in all five synthetic test environments. Tables XVIII through XXI present the results to date of PHI5-7 stress corrosion U-bend tests.

Table XXII indicates that neither AM355 or PH15-7 Mo, in the strength levels tested, are susceptible to cracking in a chlorinated hydrocarbon, a non-chlorinated hydrocarbon, or a chlorinated non-hydrocarbon.

d. Bl20VCA - The U-bend samples of this alloy that were aged for 21, 94, and 200 hours at 900 F in sealed flasks in an argon atmosphere were found to be unsuitable for test purposes.

The samples would not take the degree of plastic formation required to form a U-bend without breakage.

While every precaution was apparently taken in the preparation of these samples to avoid contamination through handling, overheating during machining, and residual air and moisture adsorption on the sample surface prior to or during aging, the effort was not successful in producing usable U-bend samples.

e. 4137 Co - During this report period, there have been a few additional failures among the various 4137 Co U-bend samples. The remaining 3 samples of the low strength group (1100°F temper or 191 Kpsi Y.S.) in chloride solution failed. Also 4 intermediate strength (232 Kpsi Y.S.) samples and 1 low strength sample failed in the sulfate environment (see Table XXV).

The high strength (550°F temper or 258 Kpsi Y.S.) specimens failed in all but the sulfide environment within a few days, the failure rate being extremely rapid in the phosphate solution.

The intermediate strength (750°F temper or 232 Kpsi Y.S.) samples failed rapidly only in the phosphate and nitrate solutions, the chloride environment being somewhat less effective in inducing failure in these specimens.

The low strength (1100°F temper or 191 Kpsi Y.S.) samples failed within a few days only in the phosphate solution, a lessening degree of effectiveness in this respect being noted for the nitrate, chloride, sulfate, and sulfide solutions.

As Table XXVI indicates the 4137 Co U-bend samples were also prone to failure in a humid atmosphere, where only the low strength samples remain after over 300 days of exposure. No failures were obtained in the laboratory and dry air environments.

f. Ardeform 30l - The cumulative data to date for these samples are given in Tables XXVII and XXVIII. Sample failures were obtained only in the chloride solution for both sets of samples. The severity of the bend for the specimens undoubtedly led to surface cracking, especially in the longitudinal samples having the outside surface in tension (see Report No. 10, p. 9, for specimen preparation). However, only the chloride solution served to promote failure, a situation similar to that shown by the AM355 test data.

Although bent beam data are not available for the Ardeform 301 alloy, we can postulate that the alloy may be stress corrosion prone in the plastically-deformed condition in a chloride environment and may not be susceptible in the stressed condition in the elastic region.

g. U-modified 4137 Co - Alloy modification of the 4137 Co alloy with about 0.8 to 1.0% uranium was found to considerably inhibit the susceptibility of the high strength samples to stress corrosion failure, as indicated in Table XXIX. Comparison of the data with the Table XXV high strength 4137 Co alloy in a chloride

solution shows that the uranium modification has effected to date a change considerably in excess of one order of magnitude in stress corrosion resistance.

During preparation of the samples for testing it was also noted that surface discoloration (oxidation or general corrosion) was considerably more rapid than for other sample materials.

Overnight exposure in the laboratory was sufficient in requiring additional surface polishing with 240-grit emery cloth.

### IV. FUTURE WORK

It is anticipated that all the remaining U-bend specimens and most of the bent beam specimens will have been subjected to varying degrees of exposure in the synthetic test environments at the end of the next report period. This will include the comparative heat samples also.

Sample preparation for coating with various paint systems and the coating phase itself should also be complete with the samples under test.

The cumulative results for the bent beam and U-bend samples now being tested will also be presented.

C. J. Owen:grr

R: 2-19-62 T: 2-19-62 Charles (C)

TABLE I

# CUMULATIVE D6Ac STRESS CORROSION BENT BEAM TESTS

Intermediate Strength (Y.S.) Samples

Test Solution	Direction of Specimen	Y.S. (.2% Offset) Kosi	Appl ied Stress* Knei	No. of	Failures to Date	Average Time to Failure,	Failure Time Range,
				200		2/200	a Lam
NaCl	Primary	206	154	9	none to 47 days	t •	;
	Secondary	223	:	;	:	:	:
NaNO <sub>3</sub>	Primary	2 06	154	9	none to 47 days	!	1
	Secondary	223		;	:	1 1	•
Na <sub>2</sub> S	Primary	206	i	;	;	!	1
:	Secondary	223	1	;	<b>5</b>	1 +	1
Na <sub>2</sub> SO <sub>4</sub>	Primary	2 06	154	9	none to 47 days	† 1	! !
l I	Secondary	223	:	!	;	1	:
NaPO	Primary	- 506	154	9	none to 47 days	i	1
•	Secondary	223	;	;	1	1	!

Specimens were austenitized at 1550°F for 20 min., oil quenched, and tempered at 840°F for 2 + 1 hrs., with an air quench after each temper. \* Samples stressed to 75% of Y.S. in holder.

TABLE II

## CUMULATIVE VASCOJET-1000 BENT BEAM TESTS

Low Strength (Y.S.) Samples

		Y.S.	Applied			Average Time	Failure Time
Test	Direction	(.2% Offset)	Stress	No. of	Failures	to Failure,	Range,
Solution	of Specimen	Kpsi	Kpsi	Specimens	to Date	days	days
NaCl	Primary	207	155	9	 •	29.7	24.5 to 36.5
	Secondary	1	!	1	1	!	;
NaNO3	Primary	207	155	9	none to 71 days	1 1	\$ 1
1	Secondary	!	:	ı	t	!	:
Na, S	Primary	207	155	9	none to 70 days	!	•
ı	Secondary	:	!	1	ı	:	:
$Na_2SO_4$	Primary	207	155	9	none to 70 days	:	:
	Secondary	1	:	1	•	;	:
$NaPO_3$	Primary	207	155	9	none to 70 days	!	i
	Secondary	<b>:</b>	i t	•		!	:

\* Samples stressed to 75% of their Y.S. in holders; all 4-inch samples.

Note: These samples were austenitized at 1900°F for 40 min., oil quenched, and tempered for Primary direction is the direction of lowest Y.S. (transverse for this alloy). 2 + 2 + 2 hrs, at 1100°F with an air quench after each temper.

TABLE III

14.13%

## CUMULATIVE VASCOJET-1000 BENT BEAM TESTS

Intermediate Strength (Y.S.) Samples

Solution	Direction of Specimen	Y.S. (,2% Offset) Kpsi	Applied Stress* Kpsi	No. of Specimens	Failures to Date	Average Time to Failure,	Range, days
Na C1	Primary Secondary	246	184	91	9 1	0.5	3 hrs. to 1 day
NaNO <sub>3</sub>	Primary Secondary	246	184	91	9 1	3.9	2.5 to 5.5
Na <sub>2</sub> S	Primary Secondary	246	184	91	none to 56 days	11	::
Na <sub>2</sub> SO <sub>4</sub>	Primary Secondary	246	184	91	<b>9</b> 1	1.5	0.5 to 3.5
Na PO 3	Primary Secondary	246	184	91	<b>9</b> 1	0.5	4 hrs. to l day

\* Samples stressed to 75% of their Y.S. in holders; all 4-inch samples.

Note: These samples were austenitized at 1900°F for 40 min., oil quenched, and tempered for Primary direction is the direction of lowest Y.S. (transverse for this alloy). 2 + 2 + 2 hrs. at 850°F with an air quench after each temper.

TABLE IV

### CUMULATIVE VASCOJET-1000 BENT BEAM TESTS

High Strength (Y.S.) Samples

Test Sylution	Direction of Specimen	Y.S. (.2% Offset) Kpsi	Applied Stress* Kpsi	No. of Specimens	Failures to Date	Average Time to Failure, days	Failure Time Range, days
Na Ci	Primary Secondary Secondary***	260 260	195 195	199	1 9 9	0.0	1 hr. to 0.5 days 2 hrs. to 0.5 days
NaNO <sub>3</sub>	Primary Secondary Secondary***	 260 260	195 195	199	199	19.0	8.5 to 25.5 8.5 to 28.5
Na <sub>2</sub> S	Primary Secondary	260	195	1 1		: :	: :
Na <sub>2</sub> SO <sub>4</sub>	Primary Secondary	260	195	19	۱ 9	1.4	0.5 to 3
NaPO <sub>3</sub>	Primary Secondary	260	195	: 9	1 9	3.3 hrs.	 1,6 hrs. to 5 hrs.

\* Samples stressed to 75% of their Y.S. in holders.

\*\* Seven-inch bent beam specimens; all others are four-inch.

Note: These samples were austenitized at 1900°F for 40 min., oil quenched, and tempered for Primary direction is the direction of lowest Y.S. (transverse for this alloy). 2 + 2 + 2 hrs, at 975°F with an air quench after each temper.

TABLE V

製物にい

# CUMULATIVE AM355 STRESS CORROSION BENT BEAM TESTS

Primary (Lowest Y.S.) Direction

		Y.S.	Applied			Average Time	Failure Time
Test	Direction	(, 2% Offset)	Stress*	No. of	Failures	to Failure,	Range,
Solution	of Specimen	Kpsi	Kpsi	Specimens	to Date	даув	days
Na Cl	Primary	224	168	4•	none to 33 days	i T	:
	Primary	235	176	4	none to 33 days	1	:
	Primary	274	205	44	none to 33 days	1	:
NaNO <sub>2</sub>	Primary	224	168	4	none to 33 days	;	•
1	Primary	235	176	4	none to 33 days	1	:
	Primary	274	205	4	none to 33 days	:	:
Na, S	Primary	224	168	:	1	1	;
ı	Primary	235	176	:	1	:	:
	Primary	274	205	1	1	1	:
Na, SO,	Primary	224	168	4	none to 33 days	;	;
# 7	Primary	235	176	4	none to 33 days	:	•
	Primary	274	205	4	none to 33 days	:	1
NaPO2	Primary	224	168	4	none to 33 days	1 1	•
)	Primary	235	176	4	none to 33 days	:	:
	Primary	274	205	4	none to 33 days	\$ 1	:

\* Samples stressed to 75% of Y.S. in holder. Sheet material rolled to strength by vendor; three Y.S. levels are 224, 235, and 274 Kpsi. Primary direction is direction of lowest Y.S. (transverse direction for this alloy).

TABLE VI

# CUMULATIVE AM355 STRESS CORROSION BENT BEAM TESTS

		Applied			Average Time	Failure Time
Test	Direction*	Stress	No. of	Failures	to Failure,	Range,
Solution	of Specimen	Kpsi	Specimens	to Date	days	days
NaCl	Secondary	180	9	none to 302 days	;	i
	Secondary	195	9	none to 286 days	:	1
	Secondary	922	9	none to 299 days	;	:
Na NO 3	Secondary	180	9	none to 302 days	;	;
)	Secondary	195	9	none to 286 days	1 1	:
	Secondary	922	9	none to 299 days	:	:
Na, S	Secondary	180	9	none to 302 days	:	;
1	Secondary	195	9	none to 286 days	!	1
	Secondary	526	9	none to 299 days	:	:
Na, SO4	Secondary	180	9	none to 302 days	;	•
1	Secondary	195	9	none to 291 days	1 1	:
	Secondary	526	9	none to 304 days	:	;
NaPO	Secondary	180	9	none to 302 days	;	i
)	Secondary	195	9	none to 291 days	t I	•
	Secondary	226	9	none to 304 days	;	;

The outer surface of the specimens is stressed to 75% of its Y.S. Note: These specimens were cold-rolled to strength level,

(250 Kpsi, 261 Kpsi, and 302 Kpsi, resp.) Above tests considered complete as of January 2, 1962.

Primary direction is direction of lowest Y.S. (transverse direction for this alloy).

TABLE VII

**袋** 

# CUMULATIVE PHI5-7 Mo STRESS CORROSION BENT BEAM TESTS

High Strength (Y.S.) Samples

	:	Y.S.	Applied		:	Average Time	Failure Time
Test Solution	Direction of Specimen	(.2% Offset) Kpsi	Stress* Kpsi	No. of Specimens	Failures to Date	to Failure, days	Range, days
NaCl	Primary	596	214	4	none to 41 days	;	ł
	Secondary	:	;	1 1	i	:	!
NaNO <sub>3</sub>	Primary	286	214	4	none to 41 days	:	;
	Secondary	!	;	1 6	1 1	;	:
Na <sub>2</sub> S	Primary	586	214	4	none to 41 days	:	;
	Secondary	!	!	i	;	:	i
Na2SO4	Primary	286	214	4	none to 41 days	1	;
	Secondary	:	!	!	:	:	1 1
Na PO3	Primary	286	214	4	none to 41 days	i	8 8
	Secondary	•	;	:	1	:	;

CR sheet material tempered 1 hr. at 900°F. Primary direction is direction of lowest Y.S. (transverse direction for this alloy). Samples stressed to 75% of Y.S. in holder.

TABLE VIII

# CUMULATIVE B120VCA STRESS CORROSION BENT BEAM TESTS

High Strength (Y.S.) Samples

198	Primary Secondary
	198

\* Samples stressed to 75% of Y.S. in holder. Solution-treated, as-received material was aged for 200 hrs. in argon at 900°F. Primary direction is direction of lowest Y.S. (transverse direction for this alloy).

TABLE IX

CUMULATIVE 4137 Co STRESS CORROSION BENT BEAM TESTS\*

Marine Exposure 190 <sup>+</sup>		Direction No. of of Specimens	No. of Specimens	Failures to Date	Average time randre time to Failure, Range, days days	Range, days
	. +	Primary	en u	ന്ധ	<b>∞</b> ~	3 to 16
	+	Primary	n 10	none to 462 days	C :	
Semi-Industrial 190		Primary	ĸ	s,	19	2 to 30
Exposure 160		Primary	2	S.	35	18 to 45
Monroeville, Pa. 130		Primary	S	none to 462 days	;	:

\* Cooperative testing program with U. S. Steel Applied Research Lab., Monroeville, Pa.

\*\* All specimens stressed to 75% of Y.S.

Note: These specimens were austenitized at 1700°F for 30 min and tempered at 550, 750 and 1100°F for 2 + 1 hrs producing Y.S. of 255, 213, and 174 Kpsi, respectively. The primary direction is direction of lowest Y.S. (transverse for this alloy).

+ 550°F Temper

++ 750°F Temper

+++ 1100°F Temper

TABLE X

# CUMULATIVE 4137 Co STRESS CORROSION BENT BEAM TESTS

Severe Industrial Exposure (Pittsburgh, Pennsylvania)

Failure Time Range, days	:::	30 to 65 	21 to 43 88 to 203
Average Time to Failure, days	111	47.2	33.6 143.2 
Failures to Date	4 to 335 days none to 335 days none to 335 days	6 4 to 335 days none to 335 days	6 5 none to 335 days
No. of Specimens	o n o	999	ം ശ
Direction of Specimen	Primary Primary Primary	Primary Primary Primary	Primary Primary Primary
Applied Stress on Specimen, Kpsi	130 <sup>+</sup> 115 ++ 95+++	195 175 140	230 200 170
% of Y.S. for Test	20	75	06

Note: These specimens were austenitized at 1700°F for 30 min and tempered at 550, 750 and 1100°F for 2 + 1 hrs producing Y.S. of 258, 232 and 191 Kpsi, respectively. The primary direction is direction of lowest Y.S.

+++ 1100°F Temper ++ 750°F Temper 550°F Temper +

TABLE XI

1. 4.16 - 14 April

## CUMULATIVE 300M STRESS CORROSION U-BEND TESTS

Primary Heat Low Strength (Y.S.) Samples

Test	Direction	Y.S. (.2% Offset)	Tensile Strength	No. of	Failures	Average Time to Failure,	Failure Time Range,
Solution	of Specimen	Kpsi	Kpei	Specimens	to Date	даув	days
NaC1	Primary	ı	•	ı	•	:	:
	Secondary	200	•	9	none to 12 days	;	:
NaNO <sub>3</sub>	Primary	1	•	ŧ	ı	;	!
•	Secondary	200	•	9	none to 12 days	!	1
Na2S	Primary	1	•	1	•	;	;
ı	Secondary	200	1	9	none to 12 days	;	:
Na <sub>2</sub> SO <sub>4</sub>	Primary	1	•	•	t	;	;
, I	Secondary	200		9	l to 12 days	;	;
Na PO <sub>3</sub>	Primary		•	•	1	;	;
	Secondary	200	•	•	9	3.0 hrs.	l to 12 hrs.

\* Y.S. estimated; tensile results not yet available.

Note: These specimens were austenitized at 1650°F for 20 min., oil quenched, and tempered at 1000°F for 2 + 2 hrs. with an air quench after each.

The primary direction is the direction of lowest Y.S. (transverse for this alloy).

TABLE XII

## CUMULATIVE 300M STRESS CORROSION U-BEND TESTS

Primary Heat Intermediate Strength (Y.S.) Samples

Test	Direction	Y.S. (.2% Offset)	Tensile Strength	No. of	Failures	Average Time to Failure,	Failure Time Range,
Solution	of Specimen	Kpsi	Крві	Specimens	to Date	days	days
NaCl	Primary	ı	ŧ	1	1	ţ	•
	Secondary	220	ı	ĸ	none to 12 days	i 1	•
NaNO2	Primary	t	•	1	1	;	ŧ
ז	Secondary	220	1	ς,	none to 12 days	!	:
Nazs	Primary	ı	•	1	ı	;	;
ı	Secondary	220	1	Ŋ	none to 12 days	:	f 1
Na, SO	Primary	1	•	1	•	;	;
<b>+</b>	Secondary	220	•	ς.	4 to 12 days	!	\$ \$
Na PO,	Primary	ı	ı	1	ı	:	;
n	Secondary	220	1	5	2	12.0 min.	9 to 19 min.

\* Y.S. estimated; tensile results not yet available.

Note: These specimens were austenitized at 1650°F for 20 min., oil quenched, and tempered at 800°F The primary direction is the direction of lowest Y.S. (transverse for this alloy). for 2 + 2 hrs. with an air quench after each.

TABLE XIII

1

### CUMULATIVE 300M STRESS CORROSION U-BEND TESTS

Primary Heat High Strength (Y.S.) Samples

in .	in. 27 hrs.	rs. ars.	^	in. to 12 hrs	n. uin.
Failure Time Range, days	1 to 30 min. 8 min. to 27 hrs.	2 to 40 hrs. 12 to 59 hrs.	1 1	4 to 16 min. 15 min. to 12 hrs	5 to 8 min. 5 to 13 min.
Average Time to Failure, days	9.2 min. ** 6.8 hrs.	15.0 hrs. 31.9 hrs.	: :	8.8 min. 2.8 hrs.	6.7 min. 9.5 min.
Failures to Date	9 9	99	none to 12 days none to 12 days	99	9
No. of Specimens	99	99	9 9	99	9 9
Tensile Strength Kpsi	i i	1 1	1 1	, ,	1 1
Y.S. (,2% Offset) Kpsi	240 250	240 250	240 250	240 250	240 250
Direction of Specimen	Primary Secondary	Primary Secondary	Primary Secondary	Primary Secondary	Primary Secondary
Test	Na Cl	NaNO <sub>3</sub>	Na <sub>2</sub> S	Na <sub>2</sub> SO <sub>4</sub>	Na PO 3

\* Y.S. estimated; tensile results not yet available.

\*\* One sample lasting 59.5 hrs. not averaged.

Note: These specimens were austenitized at 1650°F for 20 min., oil quenched, and tempered at 550°F for 2 + 2 hrs. with an air quench after each.

The primary direction is the direction of lowest Y.S. (transverse for this alloy).

### TABLE XIV

### CUMULATIVE 300M STRESS CORROSION U-BEND TESTS

Comparative Heat High Strength (Y.S.) Samples

NaCl         Primary         -	Test Solution	Direction of Specimen	Y.S. (.2% Offset) Kpsi	Tensile Strength Kpsi	No. of Specimens	Failures to Date	Average Time to Failure, days	Failure Time Range, days
Primary         - </td <td>NaC1</td> <td>Primary Secondary</td> <td>250</td> <td>1 1</td> <td>1 9</td> <td>ı <b>v</b></td> <td> 65.5 hrs.</td> <td> 40.5 50 93.5 hrs</td>	NaC1	Primary Secondary	250	1 1	1 9	ı <b>v</b>	 65.5 hrs.	 40.5 50 93.5 hrs
Primary       - </td <td>NaNO<sub>3</sub></td> <td>Primary Secondary</td> <td>250</td> <td>1 1</td> <td>1 9</td> <td>ı <b>9</b></td> <td> 26.0 hrs.</td> <td> 12 to 40 hrs.</td>	NaNO <sub>3</sub>	Primary Secondary	250	1 1	1 9	ı <b>9</b>	 26.0 hrs.	 12 to 40 hrs.
Primary       -       -       -       -       39.1 hrs.         Secondary       -<	Na <sub>2</sub> S	Primary Secondary	250	1 1	1 9	- none to 12 days	::	! !
Primary Secondary 250 - 6 6 16.2 min.	Na <sub>2</sub> SO <sub>4</sub>	Primary Secondary	250	1 1		ı 9	39.1 hrs.	12 to 58.5 hrs.
	Na PO 3	Primary Secondary	250	1 1 1	ı 9	1 9	 16.2 min.	 12 to 20 min.

\* Y.S. estimated; tensile results not yet available.

Note: These specimens were austenitized at 1650°F for 20 min., oil quenched, and tempered at 550°F for 2 + 2 hrs. with an air quench after each.

The primary direction is the direction of lowest Y.S. (transverse for this alloy).

TABLE XV

### CUMULATIVE AM355 STRESS CORROSION U-BEND TESTS

Low Strength (Y.S.) Samples

		Y.S.	Tensile			Average Time	Failure Time
Test	Direction	(.2% Offset)	Strength	No. of	Failures	to Failure,	Range,
Solution	of Specimen	Kpsi	Kpsi	Specimens	on Date	uaye	
NaCl	Primary	224	277	9	2 to 55 days	1	1
	Secondary	250	261	9	none to 134 days	82	1
NaNO <sub>2</sub>	Primary	224	277	9	none to 55 days	t t	i
1	Secondary	250	261	9	none to 134 days	i 1	•
Na, S	Primary	224	277	9	none to 55 days	1	•
1	Secondary	250	261	9	none to 134 days	1 1	1
Na, SO,	Primary	224	277	9	none to 55 days	i i	;
r 2	Secondary	250	197	9	none to 134 days		1
NaPO	Primary	224	277	9	none to 55 days	i I	i
n	Secondary	250	197	9	none to 134 days	i 1	*

These specimens were cold-rolled to strength, Note:

Primary direction is direction of lowest Y.S. (transverse direction of this alloy). Specimen thickness - .038 inches. The outer surface of the specimen is stressed beyond its Y.S.

### TABLE XVI

### CUMULATIVE AM355 STRESS CORROSION U-BEND TESTS

Intermediate Strength (Y.S.) Samples

Average Time Failure Times to Failure, Range, a days	6 6.3 0.5 to 14.5 to 133 days	3 days 3 days	3 days 3 days	days days	days 3 days
Failures to Date	6 1 to 133	none to 133 days none to 133 days	none to 133 days none to 133 days	none to 133 days none to 133 days	none to 132 days none to 133 days
No. of Specimens	9 9	9 9	99	9 9	99
Tensile Strength Kpsi	286 271	286 271	286 271	286 271	286 271
Y.S. (.2% Offset) Kpsi	235 261	235 261	235 261	235 261	235 261
Direction of Specimen	Primary Secondary	Primary Secondary	Primary Secondary	Primary Secondary	Primary Secondary
Test Solution	NaCl	NaNO <sub>3</sub>	Na <sub>2</sub> S	Na <sub>2</sub> SO <sub>4</sub>	NaPO <sub>3</sub>

Note: These specimens were cold-rolled to strength level.
The outer surface of the test specimen is stressed beyond

Primary direction is the direction of lowest Y.S. (transverse direction for this alloy). The outer surface of the test specimen is stressed beyond its Y.S.

Specimen thickness - . 0365 inches.

### TABLE XVII

THE PARTY OF

### CUMULATIVE AM355 STRESS CORROSION U-BEND TESTS

High Strength (Y.S.) Samples

		Y.S.	Tensile		7	Average Time	Failure Time
Test	Direction	(.2% Offset)	Strength	No. of	Failures	to Failure,	Range,
Solution	of Specimen	Kpsi	Kpsi	Specimens	to Date	days	days
NaCl	Primarv	274	307	9	9	3,25	0.5 to 8
	Secondary	302	311	9	1 to 133 days	;	1 1
NaNO2	Primary	274	307	9	none to 280 days	:	;
1	Secondary	302	311	9	none to 133 days	1	:
Na <sub>2</sub> S	Primary	274	307	9	none to 280 days	!	!
ı	Secondary	302	311	9	none to 133 days	:	1
Na, SO	Primary	274	307	9	none to 280 days	ł	;
# J	Secondary	302	311	9	none to 133 days	:	:
NaPO2	Primary	274	307	9	none to 280 days	†	;
1	Secondary	302	311	9	none to 133 days	!	:

Note: These specimens were cold-rolled to strength level,

The outer surface of the specimen is stressed beyond its Y.S.

Primary direction is direction of lowest Y.S. (transverse direction for this alloy).

Specimen thickness - ,033 inches.

TABLE XVIII

## CUMULATIVE PHI5-7Mo STRESS CORROSION U-BEND TESTS

Low Strength (Y.S.) Samples

		S A	Tensile			Average Time	Failure Time
÷96	Direction	(2% Offset)	Strength	No. of	Failures	to Failure,	Range,
Colution	of Specimen	Kpsi	Kpsi	Specimens	to Date	days	days
	2000						
N <sub>e</sub> Cl	Primary	201	238	9	none to 308 days	;	i
	Secondary	218	221	9	none to 308 days	:	! !
ONeN	Primary	201	238	S	none to 308 days	;	i i
3	Secondary	218	221	9	none to 308 days	;	
Sex	Primary	2 01	238	9	none to 308 days	;	;
7	Secondary	218	221	9	none to 308 days	ł	1
OSTEN	Primarv	201	238	S	none to 308 days	1	;
4~2~4	Secondary	218	221	9	none to 308 days	!	1
NaPO	Primary	201	238	ហ	none to 308 days	1	1 6
5	Secondary	218	221	9	none to 308 days	! !	:
				***************************************			

Primary direction is direction of lowest Y.S. (transverse direction for this alloy). These specimens are in the cold-rolled condition, as received. The outer surface of the specimen is stressed beyond its Y.S. Specimen thickness - 0.032 inches. Note:

TABLE XIX

## CUMULATIVE PHI5-7Mo STRESS CORROSION U-BEND TESTS

Low-Intermediate Strength (Y.S.) Samples

+ 0 0 E	Direction	Y.S.	Tensile	No. of	Failures	Average Time to Failure,	Failure Time Range,
Solution	of Specimen	Kpsi	Kpsi	Specimens	to Date	days	days
No.N	Primary	208	214	9	none to 110 days	;	i i
	Secondary	224	525	9	none to 110 days	:	1
NaNO	Primary	2.08	214	9	none to 110 days		1
6	Secondary	224	523	9	none to 110 days	!	! !
S. S.	Primary	208	214	9	none to 110 days	;	1
2721	Secondary	224	525	9	none to 110 days	t ;	i i
Na. SO.	Primary	208	214	9	none to $110$ days	;	ļ
<b>4</b> . 7	Secondary	224	529	9	none to 110 days	1 1	•
NaPO	Primary	208	214	9	none toll0 days	1	E 1
•	Secondary	224	529	9	none to 110 days	I t	i i

Note: These specimens were tempered at 1125°F for 1 hr. The outer surface of the specimens is stressed beyond its Y.S.

Primary direction is the direction of lowest Y.S. (transverse direction for this alloy).

TABLE XX

## CUMULATIVE PHI5-7Mo STRESS CORROSION U-BEND TESTS

High-Intermediate Strength (Y.S.) Samples

		Y.S.	Tensile			Average Time	Failure Time
Test	Direction	(.2% Offset)	Strength	No. of	Failures	to Failure,	Range,
Solution	of Specimen	Kpsi	Kpsi	Specimens	to Date	days	days
[C.\N	Drimary	244	254	•	none to 110 days		1
1000	t tilliat y	1 1	4 )	) ·	ייסויה בס דו מייסוי		
	Secondary	258	263	9	none to 110 days	3	1
ONeN	Primary	244	254	9	none to 110 days	1 6	;
£	Secondary	258	263	9	none to 110 days	;	1 1
Na,S	Primary	244	254	9	none to 110 days	:	!
4	Secondary	258	263	9	none to 110 days	•	;
Na, SO	Primary	244	254	9	none to 110 days	\$ 8	i
۴ ۷	Secondary	258	263	9	none to 110 days	:	* .
NaPO2	Primary	244	254	9	none to 110 days	i	:
)	Secondary	258	263	9	none to 110 days	1	!

The outer surface of the specimen is stressed beyond its Y.S. These specimens were tempered at 1075°F for 1 hr. Note:

Primary direction is direction of lowest Y.S. (transverse direction for this alloy).

TABLE XXI

## CUMULATIVE PHIS-7Mo STRESS CORROSION U-BEND TESTS

High Strength (Y.S.) Samples

Test Solution	Direction of Specimen	Y.S. (.2% Offset) Kpsi	Tensile Strength Kpsi	No. of Specimens	Failures to Date	Average Time to Failure,	Failure Time Range,
NaCl	Primary Secondary	28 <b>6</b> 289	293	9	none to 110 days	: -	0 1 1
NaNO <sub>3</sub>	Primary Secondary	286 289	293 290	9 9	none to 110 days none to 110 days	: :	; ;
Na <sub>2</sub> S	Primary Secondary	286 289	293 290	99	none to 110 days none to 110 days	11	11
Na <sub>2</sub> SO <sub>4</sub>	Primary Secondary	286 289	293 290	9 9	none to 110 days none to 110 days	! !	: :
Na PO <sub>3</sub>	Primary Secondary	286 289	293 290	9 9	none to 110 days none to 110 days	; ;	1 1

Note: These specimens were tempered at 900°F for 1 hr.

The outer surface of the specimen is stressed beyond its Y.S.

Primary direction is direction of lowest Y.S. (transverse direction for this alloy).

TABLE XXII

CUMULATIVE PHI5-7Mo AND AM355 STRESS CORROSION U-BEND TESTS

31 Kpsi	Kpsi
31 238	201
221	218
74 307	274
35 286	235
238	201
221	218
307	274
286	235
238	201
238	201
. •	274 235

Note: These specimens were cold-rolled to strength level.

The outer surface of the specimen is stressed beyond its Y.S.

Primary direction is the direction of lowest Y.S. (transverse direction for these alloys).

TABLE XXIII

1

### CUMULATIVE BI20VCA STRESS CORROSION U-BEND TESTS

Primary Heat Low Strength (Y.S.) Samples

Test	Direction of Specimen	Y.S. (,2% Offset) Kpsi	Tensile Strength Kpsi	No. of Specimens	Failures to Date	Average Time to Failure, days	Failure Time Range, days
Na C1	Primary Secondary	139 143	140 145	44	none to 99 days none to 391 days	: :	11
NaNO <sub>3</sub>	Primary Secondary	139 143	140 145	4 4	none to 99 days none to 391 days	::	11
Na <sub>2</sub> S	Primary Secondary	139 143	140 145	4 4	none to 99 days none to 99 days	1 1	: :
Na <sub>2</sub> SO <sub>4</sub>	Primary Secondary	139 143	140 145	4 4	none to 99 days none to 99 days	::	11
NaPO <sub>3</sub>	Primary Secondary	139 143	140 145	4 4	none to 99 days none to 391 days	: :	; ; ; ;

The outer surface of the specimen is stressed beyond its Y.S. The primary direction is direction of lowest Y.S. Note: These specimens are in the 'as received' condition.

TABLE XXIV

# CUMULATIVE B120VCA(HT1)\* STRESS CORROSION U-BEND TESTS

Low Strength (Y.S.) Samples

Test	Direction of Specimen	Y.S. (,2% Offset) Kpsi	Tensile Strength Kpsi	No. of Specimens	Failures to Date	Average Time to Failure, days	Failure Time Range, days
NaC1	Primary Secondary	141 144	143 148	44	none to 99 days none to 99 days	: :	::
NaNO <sub>3</sub>	Primary Secondary	141 144	143 148	4 4	none to 99 days none to 99 days	: :	11
Na <sub>2</sub> S	Primary Secondary	141 144	143 148	4 4	none to 99 days none to 99 days	; ;	: :
Na <sub>2</sub> SO <sub>4</sub>	Primary Secondary	141 144	143 148	<b>4.</b> 4.	none to 99 days none to 99 days	! !	11
Na PO 3	Primary Secondary	141 144	143 148	4 4	none to 99 days none to 99 days	! !	: :

\*HTl refers to comparative heat number 1.

The outer surface of the specimen is stressed beyond its Y.S. Note: These specimens were taken in the "as received" condition. The primary direction is direction of lowest Y.S.

TABLE XXV

The second of

### CUMULATIVE 4137 Co STRESS CORROSION U-BEND TESTS

Failure Time Range, days	0.5 to 11.5 38 to 224 367 to 532	0.5 to 1.5 29.5 to 39.5 26 to 292	:::	10 min to 2, 5 days	3 to 6 min 34 min to 5 days 0, 5 to 27 days
Average Time to Failure, days	4.3 126.0 446.0	1.2 33.8 122.6	: : :	6 - 1 -	4.5 min *0.3 days **
Failures to Date	<b>99</b> 9	999	2 to 365 days 1 to 365 days none to 365 days	6 5 to 499 days 5 to 499 days	999
No. of Specimens	999	999	r 99	9 25 9	999
Tensile Strength Kpsi	285 260 198	285 260 198	285 260 198	285 260 198	285 260 198
Y.S. (.2% Offset) Kpsi	258 232 191	258 232 191	258 232 191	258 232 191	258 232 191
Direction of Specimen	Primary Primary Primary	Primary Primary Primary	Primary Primary Primary	Primary Primary Primary	Primary Primary Primary
Test Solution	NaCl	NaNO <sub>3</sub>	Na <sub>2</sub> S	Na <sub>2</sub> SO <sub>4</sub>	Na PO 3

\* One specimen lasting 5 days not averaged.

\*\* One specimen lasting 27 days not averaged.

Note: These specimens were austenitized at 1700°F for 30 min and tempered at 550, 750 and 1100°F,

The outer surface of the specimen is stressed beyond its Y.S. The primary direction is direction of lowest Y.S. respectively for 2 + 1 hrs.

TABLE XXVI

CUMULATIVE 4137 Co STRESS CORROSION U-BEND TESTS

Test	Direction	Y.S.	Tensile	No	Hailures	Average Time	Failure Time
g	of Specimen	Kpsi	Kpsi	Specimens	to Date	days	days
Dry Air	Primary	258	285	m	none to 366 days	:	;
(dessicator)		232	260	ო	none to 366 days	:	;
	Primary	191	198	3	none to 366 days	:	•
Humid Air	Primary	258	285	4	4	6.4	4 to 11.5
(satur, with		232	260	4	4	54.5	3 to 143
water vapor)	Primary	191	198	4	none to 358 days	i i	•
Laboratory	Primary	258	285	4	none to 358 days	;	;
(exposed	Primary	232	260	4	none to 358 days	1	:
directly to lab. environ.		191	198	4	none to 358 days	<b>:</b>	;
tan. Cilvitor	<b>.</b>						

Note: These specimens were austenitized at 1700°F for 30 min and tempered at 550, 750 and 1100°F, respectively for 2 + 1 hrs.

The outer surface of the specimen is stressed beyond its Y.S.

The primary direction is direction of lowest Y.S.

TABLE XXVII

一年 できない カンス・こ

## CUMULATIVE ARDEFORM 301 STRESS CORROSION U-BEND TESTS

(Sample 1)

Direction of Specimen	Test Surface in Tension	(.2% Offset) Kpsi	Strength Kpsi	No. of Specimens	Failures to Date	Average Time to Failure, days	Failure Time Range, days
Long.	Outside (convex)	183	229	4	4	1,25	0.5 to 2.0
Long.	Inside (concave)	183	229	4	3 to 280 days	<b>:</b>	:
Trans.	Outside	<b>:</b>	•	9	none to 280 days	:	:
Long.	Outside	183	525	4.	none to 280 days	:	:
Long.	Inside	183	229	4	none to 280 days	;	1
Trans.	Outside	!	:	9	none to 280 days	;	;
Long.	Outside	183	529	4	none to 280 days	;	:
Long.	Inside	183	229	4	none to 280 days	:	;
Trans.	Outside	!	:	9	none to 280 days	•	;
Long.	Outside	183	525	4	none to 280 days	!	1
Long.	Inside	183	229	4	none to 280 days	1	;
Trans.	Outside	<b>:</b>	:	9	none to 280 days	:	•
Long.	Outside	183	525	4	none to 278 days	:	1
Long.	Inside	183	229	4	none to 278 days	;	•
Trans.	Outside		:	9	none to 278 days	:	•

Note: These specimens are in the "as received" condition.

The outer surface of the specimen is stressed beyond its Y.S.

TABLE XXVIII

## CUMULATIVE ARDEFORM 301 STRESS CORROSION U-BEND TESTS

(Sample 2)

			Y.S.	Tensile			Average Time	Failure Time
Test, Solution	Direction of Specimen	Test Surface (.2% Off in Tension Kpsi	(.2% Offset) Kpsi	Strength Kpsi	No. of Specimens	Failures to Date	to Failure, days	Range, days
NaCl	Long.	Outside (convex)		235	9	2 to 250 days	;	1
	Long.	Inside (concave)	509	235	4	2 to 250 days	;	į
	Trans.	Outside	1	;	S	none to 250 days	i	1
NaNO3	Long.	Outside	5 0 9	235	S	none to 250 days	ł	i
)	Long.	Inside	506	235	4	none to 250 days	:	:
	Trans.	Outside	i i	:	S	none to 250 days	:	!
Nazs	Long.	Outside	2 09	235	4	none to 250 days	1	! 1
1	Long.	Outside	509	235	•	none to 250 days	;	:
	Trans.	Outside	!	:	S	none to 250 days	1	ţ
Na, SO4	Long.	Outside	5 0 9	235	4	none to 250 days	ł	;
1	Long.	Inside	509	235	3	none to 250 days	;	:
	Trans.	Outside	1	1	٠ س	none to 250 days	ŀ	i
Na PO	Long.	Outside	5 09	235	4	none to 250 days	1	:
<b>n</b>	Long.	Inside	509	235	4	none to 250 days	;	į
	Trans.	Outside	!	!	2	none to 250 days	:	:

Note: These specimens are in the "as received" condition.

The outer surface of the specimen is stressed beyond the Y.S.

:

TABLE XXIX

### CUMULATIVE U-MODIFIED 4137 Co U-BEND TESTS

High Strength (Y.S.) Samples

Test Sample Solution Designation	Direction of Specimen	Tempering Temp., °F	No. of Specimens	Failures to Date	Average Time to Failure, days	Failure Time Range, days
MV-38	Secondary	400	7	none to 146 days	i I	!
MV-38	Secondary	200	2	none to 146 days	1	
MV-38	Secondary	009	7	none to 146 days	1	1
MV-40	Secondary	400	7	none to 146 days	:	
MV-40	Secondary	200	7	none to 146 days	;	:
MV-40	Secondary	009	7	none to 146 days	:	:
MV-41	Secondary	400	7	none to 146 days	!	!
MV-41	Secondary	200	7	none to 146 days	:	i
MV-41	Secondary	909	7	none to 146 days	!	!!

Note: These samples were austenitized at 1700°F for 30 min., oil quenched, and tempered at 400, 500, and 600°F for 2 + 1 hours, an air quench following each temper. The outer surface of the specimen is stressed beyond its Y.S. The primary direction is the direction of lowest Y.S. (transverse for this alloy).

### (Copies of All Reports)

		No. of Coptes
A.	Department of Defense	
	Office of the Director of Defense Research & Engineering ATTN: Mr. J. C. Barrett	
	Room 3D-1067, The Pentagon	
	Washington 25, D. C.	1
		-
	Advanced Research Project Agency	
	ATTN: Dr. G. Mock	
	The Pentagon	
	Washington 25, D. C.	1
	Commander	
	Armed Services Technical Information Agency	
	ATTN: TIPDR	
	Arlington Hall Station	
	Arlington 12, Virginia	10
	Defense Metals Information Center	
	Battelle Memorial Institute	
	Columbus, Ohio	1
	Solid Propellant Information Agency	
	Applied Physics Laboratory	
	The Johns Hopkins University	
	Silver Spring, Maryland	3
в.	Department of the Army - Technical Services	
	Office Chief of Ordnance	
	ATTN: ORDTB-Materials	
	Department of the Army	•
	Washington 25, D. C.	1
	Commanding General	•
	Aberdeen Proving Ground	
	ATTN: Dr. C. Pickett, C&CL	
	Aberdeen Proving Ground, Maryland	1
	Commendation Comment	
	Commanding General	
	Ordnance Tank-Automotive Command	
	ATTN: Mr. S. Sobak, ORDMC-IF-2	1
	Detroit 9, Michigan	1
	Commanding General	
	Ordnance Weapons Command	
	ATTN: Mr. B. Gerke, ORDOW-IA	
	Pock Taland Illinois	1

Commanding General		
U. S. Army Ballistic Missile Agency		
ATTN: Dr. G. H. Reisig	1	
Mr. P. B. Wallace, ORDAB-RPEM	1	
Documentation & Technical Information Branch	2	
ORDAB-IEE	1	
Redstone Arsenal, Alabama	•	
Commanding General		
U. S. Army Rocket & Guided Missile Agency		
ATTN: Mr. Robert Fink, ORDXR-RGS	1	
Mr. W. H. Thomas, ORDXR-IQI	1	
Redstone Arsenal, Alabama		
Commanding Officer		
Frankford Arsenal		
ATTN: Dr. H. Gisser, ORDBA-1330	1	
Mr. H. Markus, ORDBA-1320	1	
Philadelphia 37, Pa.		
Commanding Officer		
Ordnance Materials Research Office		
Watertown Arsenal		
ATTN: RPD		
Watertown 72, Mass.	1	
Commanding Officer		
Picatinny Arsenal		
· · · · · · · · · · · · · · · · · · ·	2	
ATTN: Mr. J. J. Scavuzzo, Plastics & Packaging Lab	3	
Mr. D. Stein, ORDBB-DE3 Dover, N. J.	1	
Companyly at 0.551 and		
Commanding Officer PLASTEC		
Picatinny Arsenal		
Dover, N. J.	1	
Commanding Officer		
Rock Island Arsenal		
ATTN: Materials Section, Laboratory		
Rock Island, Illinois	1	
Commanding Officer		
Springfield Armory		
ATTN: Mr. R. Korytoski, Research Materials Lab		
Springfield 1. Mass.	1	

		NO. OI	Copies
	Commanding Officer		
	Watertown Arsenal		
	ATTN: ORDBE-LX		
	Watertown 72, Mass.	3	
	Commanding Officer		
	Watervliet Arsenal		
	ATTN: Mr. F. Dashnaw, ORDBF-RR		
	Watervliet, New York	1	
	Headquarters		
	U. S. Army Signal R&D Laboratory		
	ATTN: Mr. H. H. Kedesky, SIGRA/SL-XE		
	Fort Monmouth, N. J.	1	
	•		
	Department of the Army - Other Army Agencies		
	Commander		
	Army Research Office		
	Arlingtin Hall Station		
	Arlington 12, Virginia	1	
	Chief of Research and Development		
	U. S. Army Research and Development Liaison Group		
	ATTN: Dr. B. Stein		
	APO 757, New York, N. Y.	1	
	no 757, new lock, n. 1.	•	
c.	Department of the Navy		
	Chief, Bureau of Naval Weapons		
	Department of the Navy		
	ATTN: RMMP		
	Room 2225, Munitions Building		
	Washington 25, D. C.	1	
	1001118001 13, 0. 0.	•	
	Department of the Navy		
	Office of Naval Research		
	ATTN: Code 423		
	Washington 25, D. C.	1	
	Department of the Navy		
	Special Projects Office		
	ATTN: SP 271		
	Washington 25, D. C.	1	

		NO. OI COPIE
	Commander	
	U. S. Naval Ordnance Laboratory ATTN: Code WM	
	White Oak, Silver Spring, Maryland	1
	Commander	
	U. S. Naval Ordnance Test Station	
	ATTN: Technical Library Branch	
	China Lake, California	1
	Commander	
	U. S. Naval Research Laboratory	
	ATTN: Mr. J. E. Srawley Anacostia Station	
	Washington 25, D. C.	1
D.	Department of the Air Force	
	U. S. Air Force Directorate of Research & Development	
	ATTN: Lt Col J. B. Shipp, Jr. Room 4D-313, The Pentagon	
	Washington 25, D. C.	1
	. estq	_
	Wright Air Development Division	
	ATTN: H. Zoeller, ASRCEE-1 Wright-Patterson Air Force Base, Ohio	2
	wright-Patterson Air Force Base, Onio	2
	ARDC Flight Test Center	
	ATTN: Solid Systems Division, FTRSC	e
	Edwards Air Force Base, California	5
	AMC Aeronautical Systems Center	
	ATTN: Manufacturing & Materials Technology Div, LMBMO	•
	Wright Patterson Air Force Base, Ohio	2
_		
E.	Other Government Agencies	
	National Aeronautics and Space Administration	
	ATTN: Mr. R. V. Rhode	1
	Mr. G. C. Deutsch	1
	Washington, D. C.	
	Dr. W. Lucas	
	George C. Marshall Space Flight Center	
	National Aeronautics and Space Administration ATTN: M-S&M-M	
	Huntsville, Alabama	1
		-

		NO. OI CODIES
	Dr. L. Jaffe	
	Jet Propulsion Laboratory	
	California Institute of Technology	
	4800 Oak Grove Drive	_
	Pasadena, California	1
	Mr. William A. Wilson	
	George C. Marshall Space Flight Center	
	ATTN: M-F&AE-M	
	Huntsville, Alabama	1
F.	Defense Contractors	
	Aerojet-General Corporation	
	ATTN: Librarian	
	Post Office Box 1168	
	Sacramento, California	1
	Aerojet-General Corporation	
	ATTN: Librarian	1
	Mr. C. A. Fournier	<u></u>
	Post Office Box 296	-
	Azusa, California	
	Allison Division	
	General Motors Corporation	
	ATTN: Mr. D. K. Hanink	
	Indianapolis 6, Indiana	1
	ARDE-Portland, Inc.	
	ATTN: Mr. R. Alper	
	100 Century Road	
	Paramus, N. J.	
	ratamus, N. J.	1
	Atlantic Research Corporation	
	ATTN: Mr. E. A. Olcott	
	Shirley Highway and Edsall Road	
	Alexandria, Virginia	1
	Curtiss-Wright Corporation	
	Wright Aeronautical Division	
	ATTN: Mr. R. S. Shuris	1
	Mr. A. M. Kettle, Technical Library	ī
	Wood-Ridge N I	•

	No. of Copies
Hercules Powder Company	
Allegheny Ballistics Laboratory	
ATTN: Dr. R. Steinberger	
P. O. Box 210	
Cumberland, Maryland	1
Hughes Aircraft Company	
ATTN: Librarian	
Culver City, California	1
Tapco Group	
ATTN: Mr. W. J. Piper	
23555 Euclid Avenue	
Cleveland 17, Ohio	1

(Case Matls & Fabrication)

		No. of Copies
Α.	Department of the Navy	
	Chief, Bureau of Naval Weapons Department of the Navy	
	ATTN: Mr. P. Goodwin	
	Washington 25, D. C.	1
в.	Department of the Air Force	
	Headquarters	
	Aeronautical Systems Division	
	ATTN: Dr. Tamborski, ASRCNP	•
	Wright-Patterson Air Force Base, Ohio	1
	Wright Air Development Division	
	ATTN: Mr. G. Peterson, ASRCNC-1	
	Wright-Patterson Air Force Base, Ohio	1
c.	Defense Contractors	
	Allegheny Ludlum Steel Corporation	
	Research Center	
	ATTN: Mr. R. A. Lula	
	Brackenridge, Pa.	1
	Alloyd Electronics Corporation	
	ATTN: Dr. S. S. White	
	35 Cambridge Parkway	
	Cambridge, Mass.	1
	Armco Steel Corporation	
	General Offices	
	ATTN: Mr. J. Barnett	
	Middletown, Ohio	1
	Battelle Memorial Institute	
	ATTN: Mr. R. Monroe	1
	Mr. G. Faulkner	1
	505 King Avenue	
	Columbus 1, Ohio	
	The Boeing Company	
	Aero Space Division	
	P. O. Box 3707	
	Seattle 24, Washington	1

### (Case Matls & Fabrication - Cont'd)

	No. of Copies
Borg-Warner Corporation Ingersoll Kalamazoo Division ATIN: Mr. L. E. Hershey	
1810 N. Pitcher St. Kalamazoo, Michigan	1
The Budd Company Defense Division	
ATTN: Mr. R. C. Dethloff Mr. Ohman	1 1
Philadelphia 32, Pa.	
Climax Molybdenum Company ATTN: Mr. R. R. Freeman 1270 Avenue of the Americas	
New York 20, New York	1
Douglas Aircraft Company Inc. Santa Monica Division	
ATTN: Mr. J. L. Waisman	
Santa Monica, California	1
General Electric Company Rocket Engine Section	
Flight Propulsion Laboratory Department	
Cincinnati 15, Ohio	1
Jones and Laughlin Steel Corporation ATTN: Mr. H. Steinbrenner	
45 South Montgomery Avenue	
Youngstown 1, Ohio	1
Arthur D. Little, Inc.	
ATTN: Dr. R. Davis Acorn Park	
Cambridge 40, Mass.	1
Lyon, Inc.	
ATTN: Mr. W. Martin 13881 W. Chicago Boulevard	
Detroit, Michigan	1
Manufacturing Laboratories	
ATTN: Dr. P. Fopiano	1
Dr. V. Radcliffe	1
21-35 Erie Street Cambridge 42, Mass.	

### (Case Matls & Fabrication - Cont'd)

	No. of Copies
Minneapolis-Honeywell Regulator Company 1230 Soldiers Field Road	
Boston 35, Mass.	1
Norris-Thermador Corporation ATIN: Mr. L. Shiller 5215 South Boyle Avenue Los Angeles 58, California	1
The Perkin-Elmer Corporation ATTN: Mr. H. L. Sachs Main Avenue	
Norwalk, Connecticut	1
Pratt & Whitney Aircraft ATIN: Mr. F. A. Crosby East Hartford, Connecticut	1
Reactive Metals Corporation ATIN: Mr. H. Lunderstrem Niles, Ohio	1
Republic Steel Corporation Research Center ATIN: Mr. H. P. Manger Independence, Ohio	1
Rohm & Haas Company Redstone Arsenal Research Division ATIN: Library Huntsville, Alabama	1
Space Technology Laboratories, Inc. ATTN: Technical Information Center Document Procurement P. O. Box 95001 Los Angeles 45, California	1
Thiokol Chemical Corporation Utah Division Brigham City, Utah	1
Titanium Metals Corporation ATIN: Mr. G. Erbin 233 Broadway New York N Y	1

### (Case Matls & Fabrication - Cont'd)

		No. of Copies
	Universal-Cyclops Steel Corp.	
	Stewart Street	
	Bridgeville, Pennsylvania	1
	U. S. Borax Research Corp.	
	ATTN: Mr. R. J. Brotherton	
	412 Crescent Way	•
	Anaheim, California	1
	United States Rubber Company	
	Research Center	
	ATTN: Dr. E. J. Joss	_
	Wayne, N. J.	1
D.	Educational Institutions	
	Massachusetts Institute of Technology	
	ATTN: Prof. W. A. Backofen	1
	Prof. M. C. Flemings	1
	Cambridge, Massachusetts	•
	Mellon Institute	_
	ATTN: Dr. H. L. Anthony	1
	Mr. C. J. Owen	1
	4400 Fifth Avenue	
	Pittsburgh 13, Pa.	
	Michigan State University	
	ATTN: Mr. R. N. Hammer	
	Department of Chemistry	
	East Lansing, Michigan	1
	Ohio State University	
	Research Foundation	
	ATTN: Dr. R. McMaster	
	Columbus, Ohio	1

### (Nozzle Matls & Fabrication)

		NO. Of Copies
A.	Department of the Navy	
	Chief, Bureau of Naval Weapons Department of the Navy ATTN: RRMA-22	
	Washington 25, D. C.	1
В.	Department of the Air Force	
	Wright Air Development Division ATTN: WWRCNC-2	
	Wright Patterson Air Force Base, Ohio	1
c.	Defense Contractors	
	Armour Research Foundation ATTN: Dr. N. Parrikh	
	10 West 55th Street	
	Chicago 16, Illinois	1
	onicago 10, illinois	1
	AVCO Manufacturing Corporation	
	Research and Advanced Development Division	
	201 Lowell Street	
	Wilmington, Massachusetts	1
	Beryllium Corporation	
	ATTN: Mr. W. H. Santschi	
	Reading, Pa.	1
	Carborundum Company	
	Research and Development Division	
	ATTN: Mr. C. E. Schulze	
	Niagara Falls, New York	1
	Clevite Corporation	
	Mechanical Research Division	
	ATTN: Mr. G. Davis	
	540 East 105th Street	
	Cleveland 8, Ohio	1
	Fansteel Metallurgical Corporation	
	ATTN: Mr. A. B. Michael	_
	North Chicago, Illinois	1

### (Nozzle Matls & Fabrication - Cont'd)

	No. of Copies
General Electric Company FPO Technical Information Center P. O. Box 196 Cincinnati 15, Ohio	1
High Temperature Materials, Inc. Technical Library	
31 Antwerp Street Brighton, Mass.	1
Hughes Tool Company Aircraft Division ATTN: Mr. H. Leggett Culver City, California	1
Arthur D. Little, Inc. ATTN: Dr. Flint Acorn Park Cambridge 40, Mass.	1
National Research Corporation ATTN: Mr. M. Torti 70 Memorial Drive Cambridge 42, Mass.	1
Nuclear Development Corporation of America ATTN: Dr. Oppenheimer White Plains, New York	1
Raytheon Manufacturing Company ATIN: Mr. S. D'Urso Waltham 54, Mass.	1
Solar Aircraft Company ATTN: Mr. N. B. Elsner San Diego 12, California	1
United Aircraft Corporation ATTN: M. Lubin Thoren Krass The Library 400 Main Street	
East Hartford 8, Connecticut Value Engineering Company	1
ATTN: Mr. J. Huminik Alexandria, Virginia	1

### (Nozzle Matls & Fabrication - Cont'd)

		No. of Copies
	Westinghouse Electric Corporation	
	Materials Manufacturing Department	
	ATTN: Mr. James McClure	
		1
	Blairsville, Pa.	•
	!	
•	Educational Institutions	
	Georgia Institute of Technology	
	Engineering Experiment Station	
	ATTN: Mr. J. D. Walton, Jr.	
	Ceramics Branch	
		1
	Atlanta, Georgia	•
	The Johns Hopkins University	
	Applied Physics Laboratory	
	ATTN: Dr. Avery	
	Silver Spring, Maryland	1
	oriver opring, hery tone	-
	Massachusetts Institute of Technology	
	ATTN: Dr. J. Wulff	
	Cambridge, Mass.	1